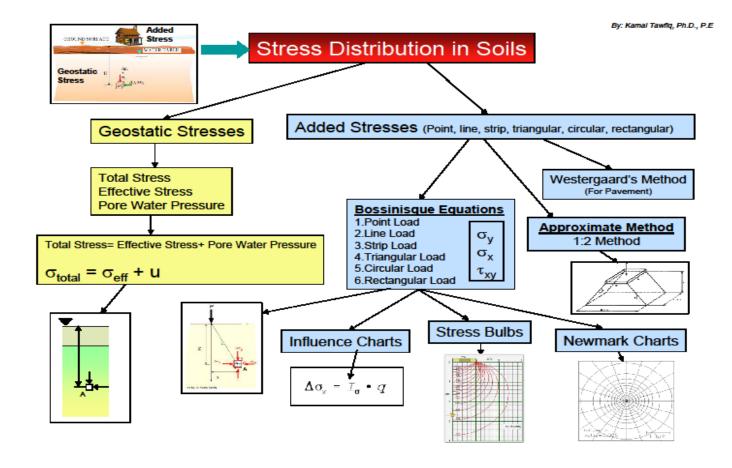
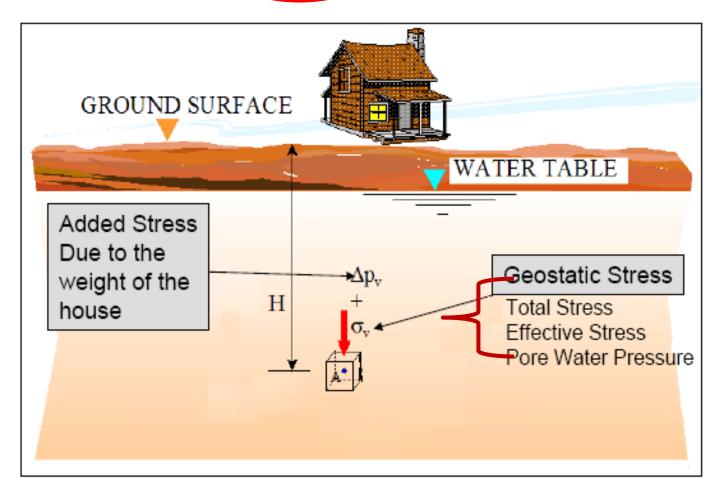
# CHAPTER 10 STRESSES IN SOIL MASS

## **TYPES OF STRESSES IN SOIL**



# **TYPES OF STRESSES IN SOIL**





# INTRODUCTION

- At a point within a soil mass, stress will be developed as a result of:
  - **The soil laying above the point (overburden)**
  - by a any structural or other loading imposed on that soil mass.

• In the preceding chapter we have discussed the stresses originated from weight of the soil itself. These stresses are called BODY STRESSES or GEOSTATIC STRESSES, or OVERBURDEN.

# INTRODUCTION

• Common examples of the external loads are as follows:

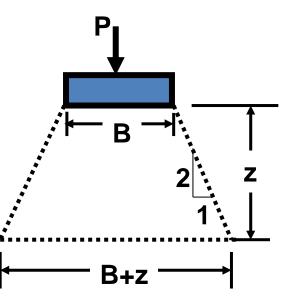
- Uniform strip loads such as the load on along wall footing of sufficient width.
- Uniformly loaded square, rectangular or circular footings such as column footings of buildings, pier footings, footings for water tanks, mats, etc

Triangular and or trapezoidal strip loads such as the loads of long earth embankments.

## **Stresses from Approximate Methods**

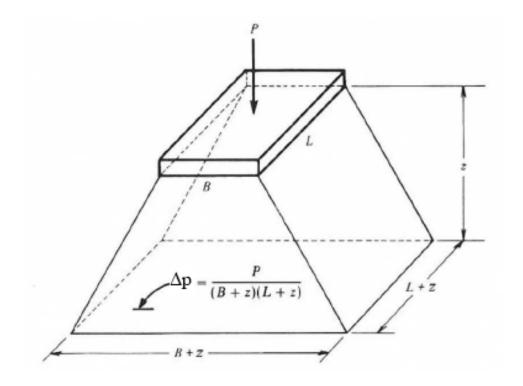
### 2:1 Method

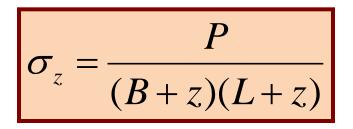
- In this method it is assumed that the STRESSED AREA is larger than the corresponding dimension of the loaded area by an amount equal to the depth of the subsurface area.
- Therefore, if a load is applied on a rectangular with dimension B and L, the stress on the soil at depth z is considered to be uniformly distributed on an area with dimension (B+z) and (L+z).



 This is called 2:1 method because the stressed area increases at a slope of 1 horizontally for each 2 of depth as measured from the depth of foundation.

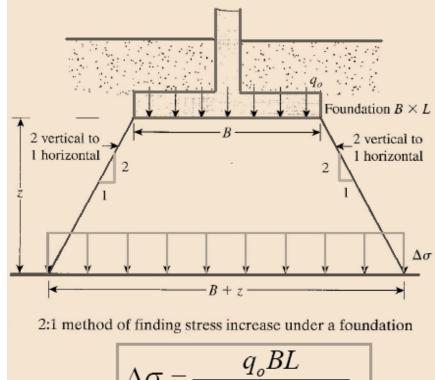
## **Stresses from Approximate Methods**





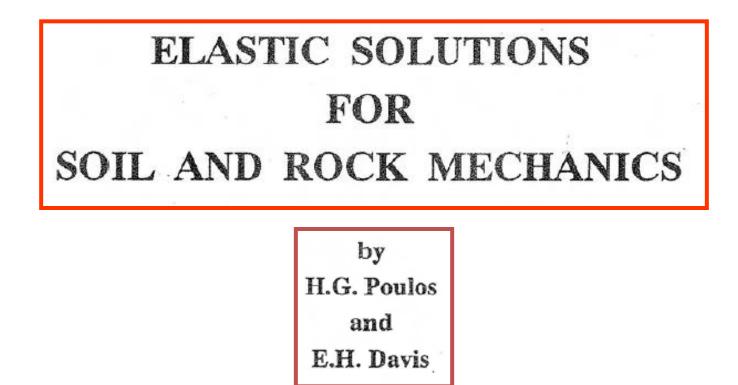
## **Stresses from Approximate Methods**

• If the load at the surface is given to be distributed, it is first converted to point load by multiplying by the area (B x L) as demonstrated in the figure below.



$$\Delta \sigma = \frac{q_o BL}{(B+z)(L+z)}$$

- There are a number of solutions which are based on the theory of elasticity. Most of them assume the following assumptions:
  - **The soil is homogeneous**
  - **4** The soil is isotropic
  - The soil is perfectly elastic infinite or semi-finite medium
- The derivations of the equations for various common loadings are tedious.
- We will concentrate only on formula, tables and charts for some of the loadings most commonly encountered in practice.
- Tens of solutions for different problems are now available in the literature. It is enough to say that a whole book <u>(Poulos and Davis)</u> is now available for the elastic solutions of various problems.



The book contains a comprehensive collection of graphs, tables and explicit solutions of problems in elasticity relevant to soil and rock mechanics.

### • The available solution depends on the following conditions:

- 1. Types of the applied load
  - 4 Point
  - Distributed
- 2. Shape of the loaded area
  - **4** Rectangular
  - Square
  - Circular
  - 🖌 etc.
- 3. Extension of the Medium
  - Half-space
  - 4 Finite
  - Iayered

- 6. Stiffness of Loaded Area Flexible Rigid
- We can see that a lot of combinations can be made from the above conditions. Next we will consider some of these solutions which are well-known and has been accepted and extensively used.

- 4. Type of soil
  - **Cohesive**
  - Cohesionless
- 5. Location of Load
  - At the surfaceAt a certain depth

Determination of vertical stress increase at a certain depth due to the application of load on the surface. The loading type includes:

- Point load
- Line load
- Uniformly distributed vertical strip load
- Linearly increasing vertical loading on a strip
- Embankment type of loading
- Uniformly loaded circular area
- Uniformly loaded rectangular area

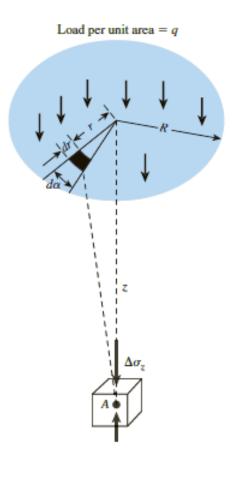
### Vertical Stress Below the Center of a Uniformly Loaded Circular Area

Using Boussinesq's solution for vertical stress  $\sigma_z$  caused by a point load one also can develop an expression for the vertical stress below the center of a uniformly loaded flexible circular area.

$$\Delta \sigma_z = q \left\{ 1 - \frac{1}{[(R/z)^2 + 1]^{3/2}} \right\}$$

#### Table 10.7 Variation of $\Delta \sigma_z/q$ with z/R [Eq. (10.27)]

z/ <b>R</b>	$\Delta \sigma_z / q$	<i>z/R</i>	$\Delta \sigma_z / q$
0	1	1.0	0.6465
0.02	0.9999	1.5	0.4240
0.05	0.9998	2.0	0.2845
0.10	0.9990	2.5	0.1996
0.2	0.9925	3.0	0.1436
0.4	0.9488	4.0	0.0869
0.5	0.9106	5.0	0.0571
0.8	0.7562		



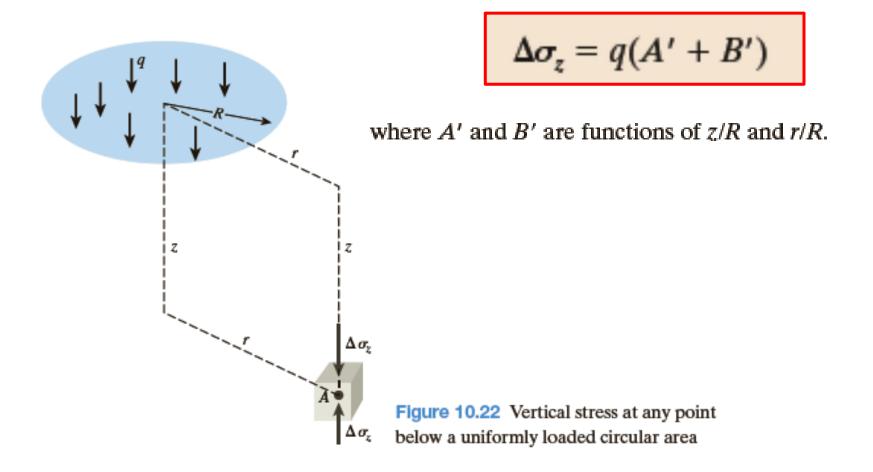


Table 10.8 Variation	of $A'$ with	$z/R$ and $r/R^*$
----------------------	--------------	-------------------

					r/R				
<i>z/</i> <b>R</b>	0	0.2	0.4	0.6	0.8	1	1.2	1.5	2
0	1.0	1.0	1.0	1.0	1.0	0.5	0	0	0
0.1	0.90050	0.89748	0.88679	0.86126	0.78797	0.43015	0.09645	0.02787	0.00856
0.2	0.80388	0.79824	0.77884	0.73483	0.63014	0.38269	0.15433	0.05251	0.01680
0.3	0.71265	0.70518	0.68316	0.62690	0.52081	0.34375	0.17964	0.07199	0.02440
0.4	0.62861	0.62015	0.59241	0.53767	0.44329	0.31048	0.18709	0.08593	0.03118
0.5	0.55279	0.54403	0.51622	0.46448	0.38390	0.28156	0.18556	0.09499	0.03701
0.6	0.48550	0.47691	0.45078	0.40427	0.33676	0.25588	0.17952	0.10010	
0.7	0.42654	0.41874	0.39491	0.35428	0.29833	0.21727	0.17124	0.10228	0.04558
0.8	0.37531	0.36832	0.34729	0.31243	0.26581	0.21297	0.16206	0.10236	
0.9	0.33104	0.32492	0.30669	0.27707	0.23832	0.19488	0.15253	0.10094	
1	0.29289	0.28763	0.27005	0.24697	0.21468	0.17868	0.14329	0.09849	0.05185
1.2	0.23178	0.22795	0.21662	0.19890	0.17626	0.15101	0.12570	0.09192	0.05260
1.5	0.16795	0.16552	0.15877	0.14804	0.13436	0.11892	0.10296	0.08048	0.05116
2	0.10557	0.10453	0.10140	0.09647	0.09011	0.08269	0.07471	0.06275	0.04496
2.5	0.07152	0.07098	0.06947	0.06698	0.06373	0.05974	0.05555	0.04880	0.03787
3	0.05132	0.05101	0.05022	0.04886	0.04707	0.04487	0.04241	0.03839	0.03150
4	0.02986	0.02976	0.02907	0.02802	0.02832	0.02749	0.02651	0.02490	0.02193
5	0.01942	0.01938				0.01835			0.01573
6	0.01361					0.01307			0.01168
7	0.01005					0.00976			0.00894
8	0.00772					0.00755			0.00703
9	0.00612					0.00600			0.00566
10								0.00477	0.00465

Table 10.8 (continued)

					r/R				
<i>z/R</i>	3	4	5	6	7	8	10	12	14
0	0	0	0	0	0	0	0	0	0
0.1	0.00211	0.00084	0.00042						
0.2	0.00419	0.00167	0.00083	0.00048	0.00030	0.00020			
0.3	0.00622	0.00250							
0.4									
0.5	0.01013	0.00407	0.00209	0.00118	0.00071	0.00053	0.00025	0.00014	0.00009
0.6									
0.7									
0.8									
0.9									
1	0.01742	0.00761	0.00393	0.00226	0.00143	0.00097	0.00050	0.00029	0.00018
1.2	0.01935	0.00871	0.00459	0.00269	0.00171	0.00115			
1.5	0.02142	0.01013	0.00548	0.00325	0.00210	0.00141	0.00073	0.00043	0.00027
2	0.02221	0.01160	0.00659	0.00399	0.00264	0.00180	0.00094	0.00056	0.00036
2.5	0.02143	0.01221	0.00732	0.00463	0.00308	0.00214	0.00115	0.00068	0.00043
3	0.01980	0.01220	0.00770	0.00505	0.00346	0.00242	0.00132	0.00079	0.00051
4	0.01592	0.01109	0.00768	0.00536	0.00384	0.00282	0.00160	0.00099	0.00065
5	0.01249	0.00949	0.00708	0.00527	0.00394	0.00298	0.00179	0.00113	0.00075
6	0.00983	0.00795	0.00628	0.00492	0.00384	0.00299	0.00188	0.00124	0.00084
7	0.00784	0.00661	0.00548	0.00445	0.00360	0.00291	0.00193	0.00130	0.00091
8	0.00635	0.00554	0.00472	0.00398	0.00332	0.00276	0.00189	0.00134	0.00094
9	0.00520	0.00466	0.00409	0.00353	0.00301	0.00256	0.00184	0.00133	0.00096
10	0.00438	0.00397	0.00352	0.00326	0.00273	0.00241			

					r/R				
/R	0	0.2	0.4	0.6	0.8	1	1.2	1.5	2
0	0	0	0	0	0	0	0	0	0
0.1	0.09852	0.10140	0.11138	0.13424	0.18796	0.05388	-0.07899	-0.02672	-0.00845
0.2	0.18857	0.19306	0.20772	0.23524	0.25983	0.08513	-0.07759	-0.04448	-0.01593
0.3	0.26362	0.26787	0.28018	0.29483	0.27257	0.10757	-0.04316	-0.04999	-0.02166
0.4	0.32016	0.32259	0.32748	0.32273	0.26925	0.12404	-0.00766	-0.04535	-0.02522
0.5	0.35777	0.35752	0.35323	0.33106	0.26236	0.13591	0.02165	-0.03455	-0.02651
0.6	0.37831	0.37531	0.36308	0.32822	0.25411	0.14440	0.04457	-0.02101	
0.7	0.38487	0.37962	0.36072	0.31929	0.24638	0.14986	0.06209	-0.00702	-0.02329
0.8	0.38091	0.37408	0.35133	0.30699	0.23779	0.15292	0.07530	0.00614	
0.9	0.36962	0.36275	0.33734	0.29299	0.22891	0.15404	0.08507	0.01795	
1	0.35355	0.34553	0.32075	0.27819	0.21978	0.15355	0.09210	0.02814	-0.01005
1.2	0.31485	0.30730	0.28481	0.24836	0.20113	0.14915	0.10002	0.04378	0.00023
1.5	0.25602	0.25025	0.23338	0.20694	0.17368	0.13732	0.10193	0.05745	0.01385
2	0.17889	0.18144	0.16644	0.15198	0.13375	0.11331	0.09254	0.06371	0.02836
2.5	0.12807	0.12633	0.12126	0.11327	0.10298	0.09130	0.07869	0.06022	0.03429
3	0.09487	0.09394	0.09099	0.08635	0.08033	0.07325	0.06551	0.05354	0.03511
4	0.05707	0.05666	0.05562	0.05383	0.05145	0.04773	0.04532	0.03995	0.03066
5	0.03772	0.03760				0.03384			0.02474
6	0.02666					0.02468			0.01968
7	0.01980					0.01868			0.01577
8	0.01526					0.01459			0.01279
9	0.01212					0.01170			0.01054
0								0.00924	0.00879

					r/R			r/R									
z/R	3	4	5	6	7	8	10	12	14								
0	0	0	0	0	0	0	0	0	0								
0.1	-0.00210	-0.00084	-0.00042														
0.2	-0.00412	-0.00166	-0.00083	-0.00024	-0.00015	-0.00010											
0.3	-0.00599	-0.00245															
0.4																	
0.5	-0.00991	-0.00388	-0.00199	-0.00116	-0.00073	-0.00049	-0.00025	-0.00014	-0.00009								
0.6																	
0.7																	
0.8																	
0.9																	
1	-0.01115	-0.00608	-0.00344	-0.00210	-0.00135	-0.00092	-0.00048	-0.00028	-0.00018								
1.2	-0.00995	-0.00632	-0.00378	-0.00236	-0.00156	-0.00107											
1.5	-0.00669	-0.00600	-0.00401	-0.00265	-0.00181	-0.00126	-0.00068	-0.00040	-0.00026								
2	0.00028	0.00410	0.00371	0.00278	0.00202	0.00148	0.00084	0.00050	0.00033								
2.5	0.00661	-0.00130	-0.00271	-0.00250	-0.00201	-0.00156	-0.00094	-0.00059	-0.00039								
3	0.01112	0.00157	-0.00134	-0.00192	-0.00179	-0.00151	-0.00099	-0.00065	-0.00046								
4	0.01515	0.00595	0.00155	-0.00029	-0.00094	-0.00109	-0.00094	-0.00068	-0.00050								
5	0.01522	0.00810	0.00371	0.00132	0.00013	-0.00043	-0.00070	-0.00061	-0.00049								
6	0.01380	0.00867	0.00496	0.00254	0.00110	0.00028	-0.00037	-0.00047	-0.00045								
7	0.01204	0.00842	0.00547	0.00332	0.00185	0.00093	-0.00002	-0.00029	-0.00037								
8	0.01034	0.00779	0.00554	0.00372	0.00236	0.00141	0.00035	0.00008	0.00025								
9	0.00888	0.00705	0.00533	0.00386	0.00265	0.00178	0.00066	0.00012	-0.00012								
10	0.00764	0.00631	0.00501	0.00382	0.00281	0.00199											

	-	100.000	the first of		-(D	0	0.7	0.4	0.6
z/R	0	0.2	0.4	0.6	z/R		0.2	0.4	0.6
0	1.0	1.0	1.0	1.0	0	0	0	0	0
0.1	0.90050	0.89748	0.88679	0.86126	0.1	0.09852	0.10140	0.11138	0.13424
0.2	0.80388	0.79824	0.77884	0.73483	0.2	0.18857	0.19306	0.20772	0.23524
0.2	0.71265	0.79824	0.68316	0.62690	0.3	0.26362	0.26787	0.28018	0.29483
0.4	0.62861	0.62015	0.59241	0.53767	0.4	0.32016	0.32259	0.32748	0.32273
0.5	0.55279	0.54403	0.51622	0.46448	0.5	0.35777	0.35752	0.35323	0.33106
0.6	0.48550	0.47691	0.45078	0.40447	0.6	0.37831	0.37531	0.36308	0.32822
0.7	0.42654	0.41874	0.39491	0.35428	0.7	0.38487	0.37962	0.36072	0.31929
0.8	0.37531	0.36832	0.34729	0.31243	0.8	0.38091	0.37408	0.35133	0.30699
0.9	0.33104	0.32492	0.30669	0.27707	0.9	0.36962	0.36275	0.33734	0.2929
1	0.29289	0.28763	0.27005	0.24697	1	0.35355	0.34553	0.32075	0.27819
1.2	0.23178	0.22795	0.21662	0.19890	1.2	0.31485	0.30730	0.28481	0.2483
1.5	0.16795	0.16552	0.15877	0.14804	1.5	0.25602	0.25025	0.23338	0.2069
2	0.10557	0.10453	0.10140	0.09647	2	0.17889	0.18144	0.16644	0.15198
2.5	0.07152	0.07098	0.06947	0.06698	2.5	0.12807	0.12633	0.12126	0.1132
3	0.05132	0.05101	0.05022	0.04886	3	0.09487	0.09394	0.09099	0.0863
4	0.02986	0.02976	0.02907	0.02802	4	0.05707	0.05666	0.05562	0.05383
5	0.01942	0.01938			5	0.03772	0.03760	11111111	0.00000

Table 10.7 Variation of  $\Delta \sigma_z/q$  with z/R [Eq. (10.27)]

z/ <b>R</b>	$\Delta \sigma_z / q$	z/ <b>R</b>	$\Delta \sigma_z / q$
0	1	1.0	0.6465
0.02	0.9999	1.5	0.4240
0.05	0.9998	2.0	0.2845
0.10	0.9990	2.5	0.1996
0.2	0.9925	3.0	0.1436
0.4	0.9488	4.0	0.0869
0.5	0.9106	5.0	0.0571
0.8	0.7562		

$$\Delta \sigma_z = q(A' + B')$$

Below the center r/R = 0 For z/R=0.2 A'=0.80388 (Table 10.8), B'= 0.18857 (Table 10.9) I= 0.80388 +0.18857= 0.99245 same as from Table 10.7

# **EXAMPLE 10.12**

#### Example 10.12

Consider a uniformly loaded flexible circular area on the ground surface, as shown in Fig. 10.22. Given: R = 3 m and uniform load q = 100 kN/m<sup>2</sup>.

Calculate the increase in vertical stress at depths of 1.5, 3, 4.5, 6, and 12 m below the ground surface for points at (a) r = 0 and (b) r = 4.5 m.

#### Solution

From Eq. (10.28),

$$\Delta \sigma_z = q \left( A' + B' \right)$$

Given R = 3 m and q = 100 kN/m<sup>2</sup>.

#### Part a

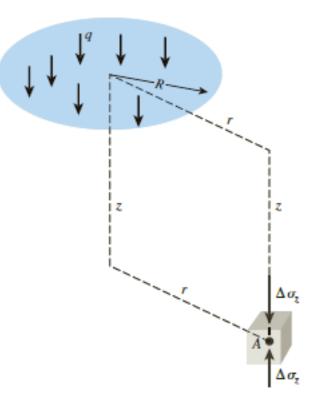
We can prepare the following table. (*Note:* r/R = 0. A' and B' values are from Tables 10.8 and 10.9.)

Depth, z (m)	z/R	A'	B'	$\Delta\sigma_z$ (kN/m <sup>2</sup> )
1.5	0.5	0.553	0.358	91.1
3	1.0	0.293	0.354	64.7
4.5	1.5	0.168	0.256	42.4
6	2.0	0.106	0.179	28.5
12	4.0	0.03	0.057	8.7

#### Part b

T/R = 4.3/3 = 1.3										
Depth, z (m)	z/R	A'	B'	$\Delta\sigma_z$ (kN/m <sup>2</sup> )						
1.5	0.5	0.095	0.035	6.0						
3	1.0	0.098	0.028	12.6						
4.5	1.5	0.08	0.057	13.7						
6	2.0	0.063	0.064	12.7						
12	4.0	0.025	0.04	6.5						

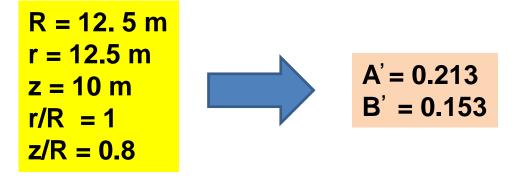
r/D = 4.5/2 = 1.5



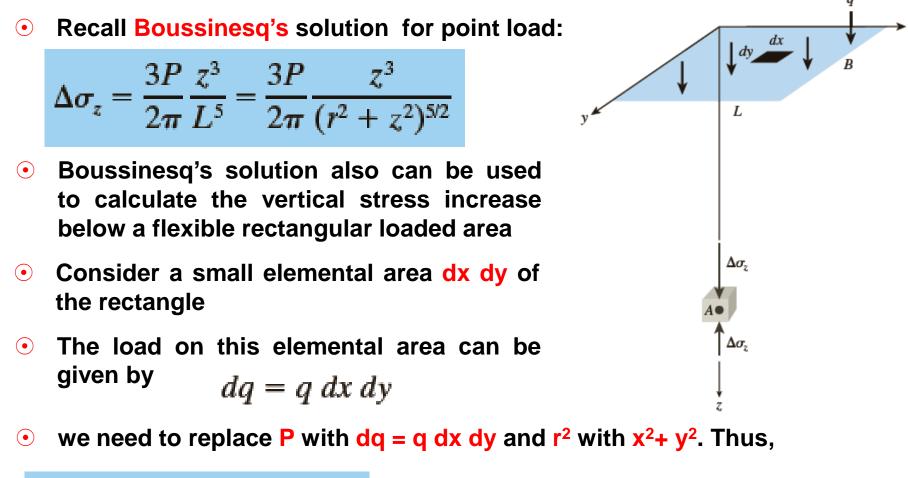


Circular tank, 25 m diameter with bearing pressure q = 122 kPa. Find stress induced by the tank 10 m below the <u>edge.</u>

$$\Delta\sigma_z = q(A'+B')$$



Δσ<sub>z</sub> = (0.213+0.153) x 122 = 44.65 kPa



### **Corner of the rectangular area**

• The increase in the stress, at point A caused by the entire loaded area can now be determined by integrating the preceding equation. We obtain

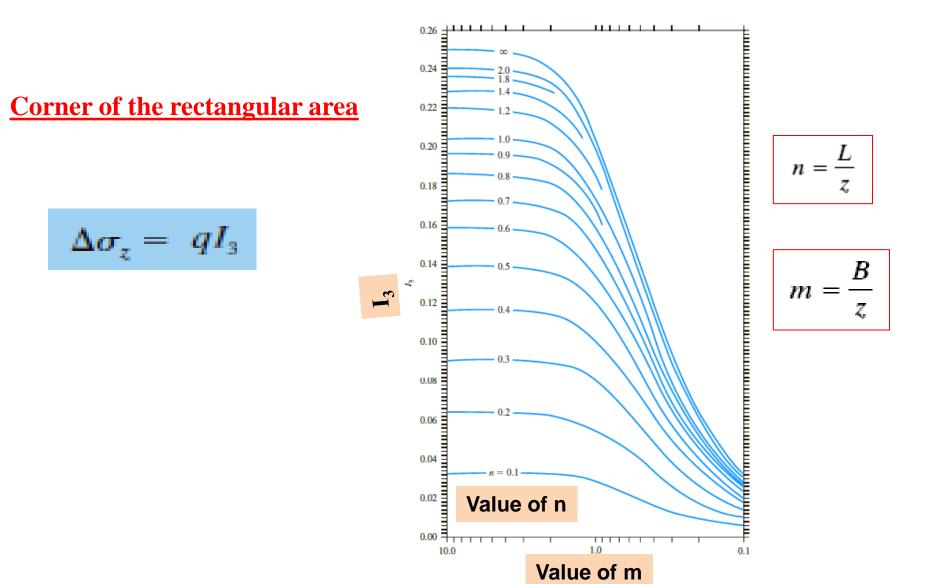
$$\Delta \sigma_{z} = \int d\sigma_{z} = \int_{y=0}^{B} \int_{x=0}^{L} \frac{3qz^{3}(dx \, dy)}{2\pi(x^{2} + y^{2} + z^{2})^{5/2}} = qI_{3}$$

$$I_{3} = \frac{1}{4\pi} \left[ \frac{2mn\sqrt{m^{2} + n^{2} + 1}}{m^{2} + n^{2} + m^{2}n^{2} + 1} \left( \frac{m^{2} + n^{2} + 2}{m^{2} + n^{2} + 1} \right) \frac{\text{Eq. 10.32}}{m^{2} + n^{2} + 1} + \tan^{-1} \left( \frac{2mn\sqrt{m^{2} + n^{2} + 1}}{m^{2} + n^{2} - m^{2}n^{2} + 1} \right) \right]$$

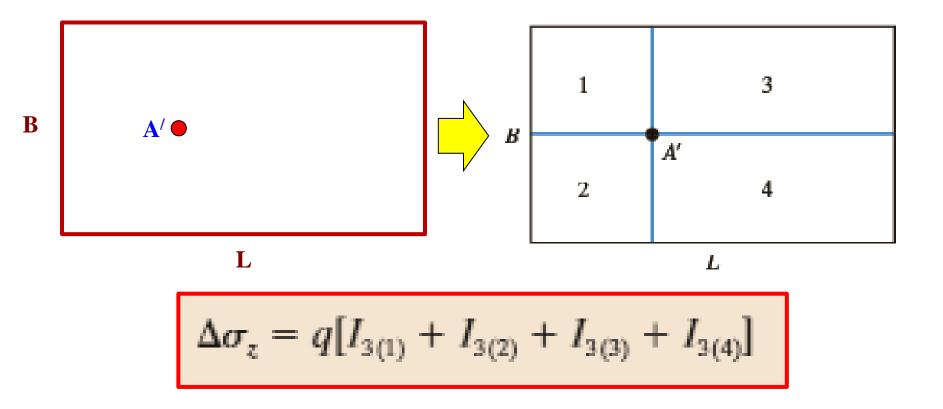
$$m = \frac{B}{z}$$

#### Table 10.10 Variation of $I_3$ with m and n [Eq. (10.32)]

					т					
n	0.1	0.2	0.3	0.4	0.5	0.6	0.7	0.8	0.9	1.0
0.1	0.0047	0.0092	0.0132	0.0168	0.0198	0.0222	0.0242	0.0258	0.0270	0.0279
0.2	0.0092	0.0179	0.0259	0.0328	0.0387	0.0435	0.0474	0.0504	0.0528	0.0547
0.3	0.0132	0.0259	0.0374	0.0474	0.0559	0.0629	0.0686	0.0731	0.0766	0.0794
0.4	0.0168	0.0328	0.0474	0.0602	0.0711	0.0801	0.0873	0.0931	0.0977	0.1013
0.5	0.0198	0.0387	0.0559	0.0711	0.0840	0.0947	0.1034	0.1104	0.1158	0.1202
0.6	0.0222	0.0435	0.0629	0.0801	0.0947	0.1069	0.1168	0.1247	0.1311	0.1361
0.7	0.0242	0.0474	0.0686	0.0873	0.1034	0.1169	0.1277	0.1365	0.1436	0.1491
0.8	0.0258	0.0504	0.0731	0.0931	0.1104	0.1247	0.1365	0.1461	0.1537	0.1598
0.9	0.0270	0.0528	0.0766	0.0977	0.1158	0.1311	0.1436	0.1537	0.1619	0.1684
1.0	0.0279	0.0547	0.0794	0.1013	0.1202	0.1361	0.1491	0.1598	0.1684	0.1752
1.2	0.0293	0.0573	0.0832	0.1063	0.1263	0.1431	0.1570	0.1684	0.1777	0.1851
1.4	0.0301	0.0589	0.0856	0.1094	0.1300	0.1475	0.1620	0.1739	0.1836	0.1914
1.6	0.0306	0.0599	0.0871	0.1114	0.1324	0.1503	0.1652	0.1774	0.1874	0.1955
1.8	0.0309	0.0606	0.0880	0.1126	0.1340	0.1521	0.1672	0.1797	0.1899	0.1981
2.0	0.0311	0.0610	0.0887	0.1134	0.1350	0.1533	0.1686	0.1812	0.1915	0.1999
2.5	0.0314	0.0616	0.0895	0.1145	0.1363	0.1548	0.1704	0.1832	0.1938	0.2024
3.0	0.0315	0.0618	0.0898	0.1150	0.1368	0.1555	0.1711	0.1841	0.1947	0.2034
4.0	0.0316	0.0619	0.0901	0.1153	0.1372	0.1560	0.1717	0.1847	0.1954	0.2042
5.0	0.0316	0.0620	0.0901	0.1154	0.1374	0.1561	0.1719	0.1849	0.1956	0.2044
6.0	0.0316	0.0620	0.0902	0.1154	0.1374	0.1562	0.1719	0.1850	0.1957	0.2045



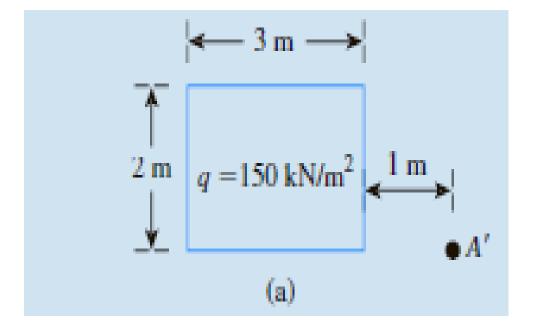
<u>The increase in the stress at any point below a rectangularly loaded</u> <u>area</u>



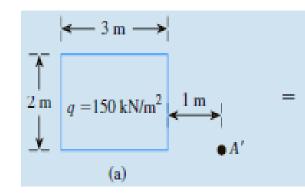


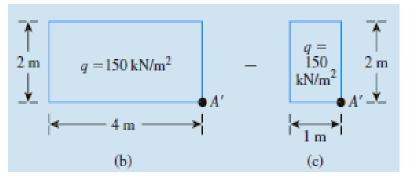
### Example 10.13

The plan of a uniformly loaded rectangular area is shown in Figure 10.27a. Determine the vertical stress increase  $\Delta \sigma_z$  below point A' at a depth of z = 4 m.



# **EXAMPLE 10.13**





#### Solution

The stress increase  $\Delta \sigma_{z}$  can be written as

$$\Delta \sigma_z = \Delta \sigma_{z(1)} - \Delta \sigma_{z(2)}$$

where

 $\Delta \sigma_{z(1)} =$  stress increase due to the loaded area shown in Figure 10.27b  $\Delta \sigma_{z(2)} =$  stress increase due to the loaded area shown in Figure 10.27c

For the loaded area shown in Figure 10.27b:

$$m = \frac{B}{z} = \frac{2}{4} = 0.5$$
$$n = \frac{L}{z} = \frac{4}{4} = 1$$

From Figure 10.24 for m = 0.5 and n = 1, the value of  $I_3 = 0.1225$ . So

$$\Delta \sigma_{z(1)} = qI_3 = (150)(0.1225) = 18.38 \text{ kN/m}$$

Similarly, for the loaded area shown in Figure 10.27c:

$$m = \frac{B}{z} = \frac{1}{4} = 0.25$$
$$n = \frac{L}{z} = \frac{2}{4} = 0.5$$

Thus,  $I_3 = 0.0473$ . Hence,

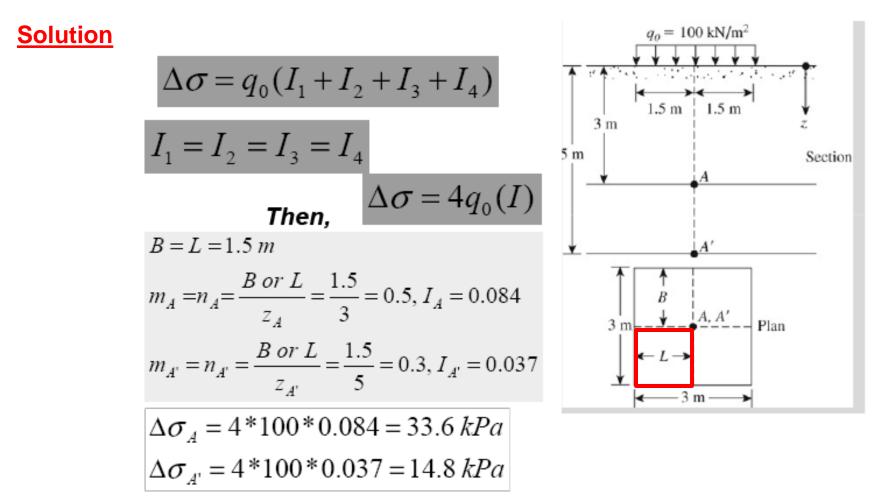
$$\Delta \sigma_{z(2)} = (150)(0.0473) = 7.1 \text{ kN/m}^2$$

So

$$\Delta \sigma_z = \Delta \sigma_{z(1)} - \Delta \sigma_{z(2)} = 18.38 - 7.1 = 11.28 \text{ kN/m}^2$$



Determine the increase in stress at point A and A' below the footing shown below.



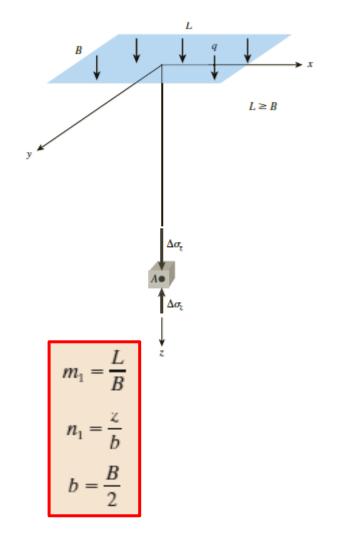
### <u>Vertical stress increase below the center of a rectangular area</u>

$$\Delta \sigma_z = q I_4$$

$$I_4 = \frac{2}{\pi} \left[ \frac{m_1 n_1}{\sqrt{1 + m_1^2 + n_1^2}} \frac{1 + m_1^2 + 2n_1^2}{(1 + n_1^2)(m_1^2 + n_1^2)} + \sin^{-1} \frac{m_1}{\sqrt{m_1^2 + n_1^2}\sqrt{1 + n_1^2}} \right]$$

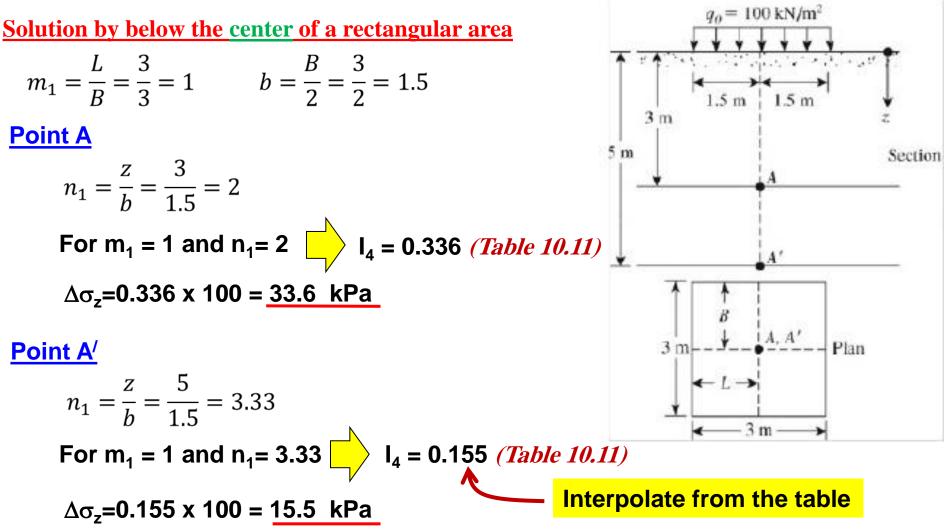
#### Table 10.11 Variation of $I_4$ with $m_1$ and $n_1$ [Eq. (10.37)]

1 2	3	4		I <sub>1</sub>														
	3	4	-				<i>m</i> <sub>1</sub>											
			5	6	7	8	9	10										
994 0.997	0.997	0.997	0.997	0.997	0.997	0.997	0.997	0.997										
960 0.976	0.977	0.977	0.977	0.977	0.977	0.977	0.977	0.977										
892 0.932	0.936	0.936	0.937	0.937	0.937	0.937	0.937	0.937										
800 0.870	0.878	0.880	0.881	0.881	0.881	0.881	0.881	0.881										
701 0.800	0.814	0.817	0.818	0.818	0.818	0.818	0.818	0.818										
606 0.727	0.748	0.753	0.754	0.755	0.755	0.755	0.755	0.755										
522 0.658	0.685	0.692	0.694	0.695	0.695	0.696	0.696	0.696										
449 0.593	0.627	0.636	0.639	0.640	0.641	0.641	0.641	0.642										
388 0.534	0.573	0.585	0.590	0.591	0.592	0.592	0.593	0.593										
336 0.481	0.525	0.540	0.545	0.547	0.548	0.549	0.549	0.549										
179 0.293	0.348	0.373	0.384	0.389	0.392	0.393	0.394	0.395										
108 0.190	0.241	0.269	0.285	0.293	0.298	0.301	0.302	0.303										
072 0.131	0.174	0.202	0.219	0.229	0.236	0.240	0.242	0.244										
051 0.095	0.130	0.155	0.172	0.184	0.192	0.197	0.200	0.202										
038 0.072	0.100	0.122	0.139	0.150	0.158	0.164	0.168	0.171										
029 0.056	0.079	0.098	0.113	0.125	0.133	0.139	0.144	0.147										
023 0.045	0.064	0.081	0.094	0.105	0.113	0.119	0.124	0.128										
019 0.037	0.053	0.067	0.079	0.089	0.097	0.103	0.108	0.112										
	960         0.976           892         0.932           800         0.870           701         0.800           506         0.727           522         0.658           149         0.593           888         0.534           336         0.481           179         0.293           108         0.190           072         0.131           051         0.095           038         0.072           029         0.056	960         0.976         0.977           892         0.932         0.936           800         0.870         0.878           701         0.800         0.814           506         0.727         0.748           522         0.658         0.685           149         0.593         0.627           888         0.534         0.573           336         0.481         0.525           179         0.293         0.348           108         0.190         0.241           172         0.131         0.174           151         0.095         0.130           138         0.072         0.100           129         0.056         0.079           123         0.045         0.064	060         0.976         0.977         0.977           082         0.932         0.936         0.936           080         0.870         0.878         0.880           701         0.800         0.814         0.817           506         0.727         0.748         0.753           522         0.658         0.685         0.692           149         0.593         0.627         0.636           588         0.534         0.573         0.585           336         0.481         0.525         0.540           179         0.293         0.348         0.373           108         0.190         0.241         0.269           172         0.131         0.174         0.202           1051         0.095         0.130         0.155           108         0.072         0.100         0.122           109         0.056         0.079         0.098           1023         0.045         0.064         0.081	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	$\begin{array}{cccccccccccccccccccccccccccccccccccc$										





Determine the increase in stress at point A and A' below the footing shown below.



# EXAMPLE

For the flexible footing shown below, determine the increase in the vertical stress at depth of z = 5 below point C for the uniformly distributed surface load q.

### **Solution**

To solve the problem, expand the footing to reach the point C. The "new" footing is a 13 by 5. The influence value  $I_3$  is found for this condition,

m = B/z = 5/5 = 1n = L/z = 13/5 = 2.6

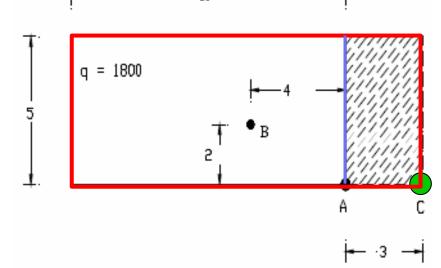
therefore  $I_3 = 0.200$ 

Now the shaded expanded area of  $3 \times 5$  is also analyzed and then subtracted from the previous result.

m = B/z = 3/5 = 0.6n = L/z = 5/5 = 1

therefore  $I_3' = 0.137$ 

Therefore  $\Delta p = q(I_3 - I_3') = (1800)(0.200 - 0.137) = 117$ 



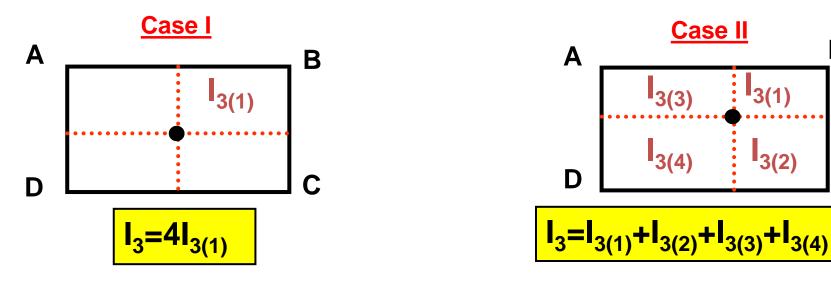
10

**Some Possible Cases** 

Β

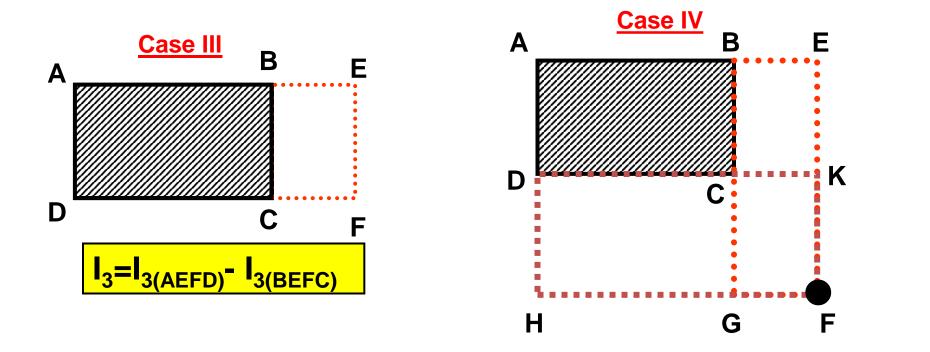
С

### Loaded area: ABCD



**Some Possible Cases** 

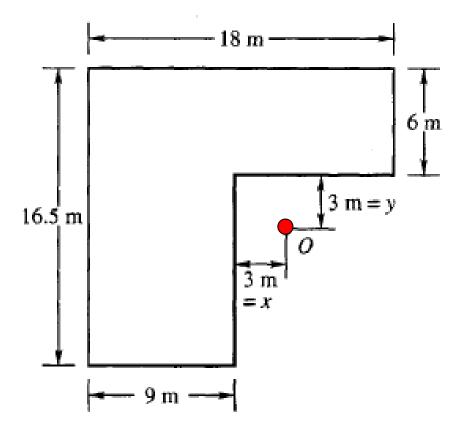
### Loaded area: ABCD



I<sub>3</sub>=I<sub>3(HAEF)</sub> - I<sub>3(GBEF)</sub> -I<sub>3(HDKF)</sub>+I<sub>3(GCKF)</sub>

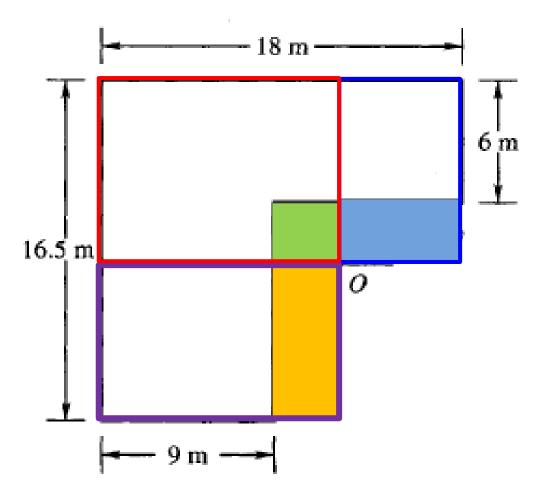
## EXAMPLE

A raft foundation of the size given below carries a uniformly distributed load of  $300 \text{ kN/m^3}$ . Estimate the vertical pressure at a depth 9 m below point O marked in the figure.

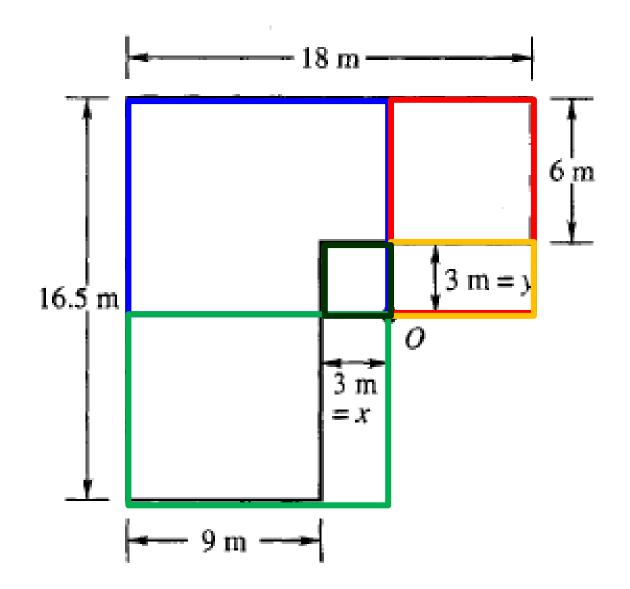




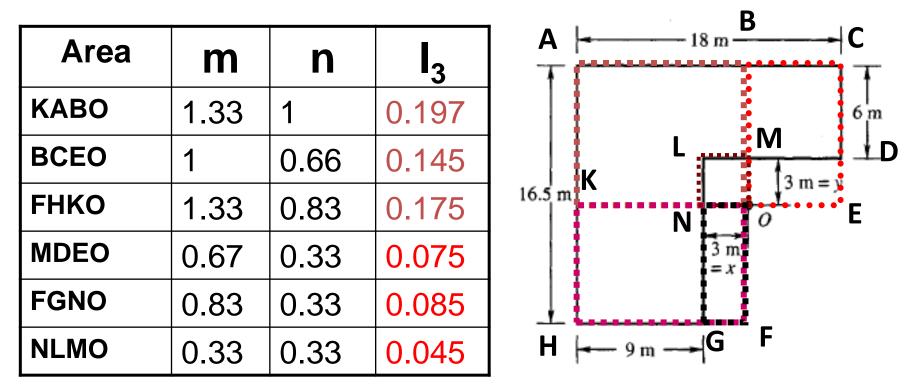
## **Approach 1: Superposition**







#### Loaded area is ACDLGH



I = 0.197 + .145 + .175 - .075 - .085 - .045 = 0.312

 $\Delta \sigma = 0.312 \times 300 = \underline{93.6} \text{ kN/m}^2$ 

## **Newmark's Influence Chart**

- Newmark (1942) constructed an influence chart based on the Boussinesq's solution.
- This chart can be used to determine the <u>vertical</u> stress at <u>any point</u> below <u>uniformly loaded flexible</u> area of <u>any shape</u>.
- Using the value of (R/z) obtained from Eq. (\*) for various pressure ratios (i.e  $\Delta \sigma_z/q$ ), Newmark (1942) presented an influence chart that can be used to determine the vertical pressure at any point below a uniformly loaded flexible area of any shape.

$\Delta \sigma J q$	R/z	$\Delta \sigma_z / q$	R/z
0	0	0.55	0.8384
0.05	0.1865	0.60	0.9176
0.10	0.2698	0.65	1.0067
0.15	0.3383	0.70	1.1097
0.20	0.4005	0.75	1.2328
0.25	0.4598	0.80	1.3871
0.30	0.5181	0.85	1.5943
0.35	0.5768	0.90	1.9084
0.40	0.6370	0.95	2.5232
0.45	0.6997	1.00	00
0.50	0.7664		

$$P_{z} = \sqrt{\left(1 - \frac{\Delta \sigma_{z}}{q}\right)^{-\frac{2}{3}} - 1}$$
 (\*)

## **Newmark's Influence Chart**

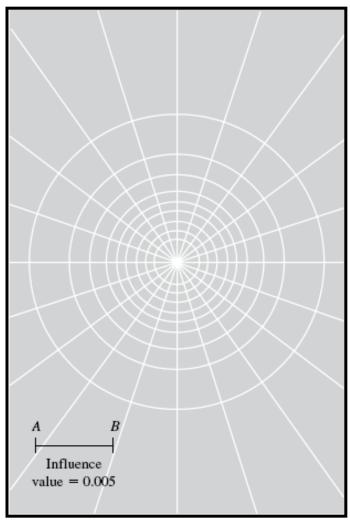
• The radii of the circles are equal to (R/z) values corresponding to

$$\Delta \sigma_z / q = 0, 0.1, 0.2, \dots 1$$

Note:

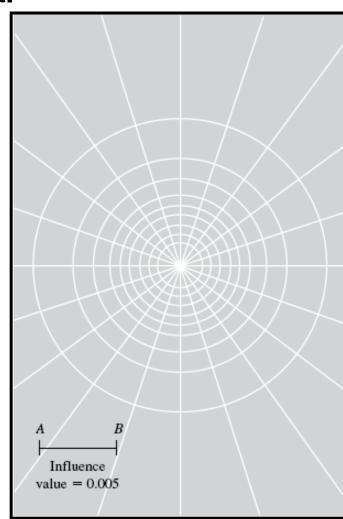
For  $\Delta \sigma_z/q = 0$ , R/z = 0, and for  $\sigma_z/q = 1$ ,  $R/z = \infty$ , so <u>nine</u> circles are shown

- The unit length for plotting the circle is  $\overline{AB}$
- The circles are divided by equally spaced radial lines



## **Newmark's Influence Chart**

- The influence value of the chart is given by 1/N, where N is equal to the number of elements in the chart.
- In the shown chart, there are 200 elements; hence the influence value is 0.005.
- The area of each segment represents an equal proportion of the applied surface stress at a depth z below the surface.



# **Procedures for Using the Chart**

The procedure for obtaining vertical pressure at any point below a loaded area is as follows:

- 1. Determine the depth *z* below the uniformly loaded area at which the stress increase is required.
- 2. Plot the plan of the loaded area with a scale of z equal to the unit length of the chart  $(\overline{AB})$ .
- Place the plan (plotted in step 2) on the influence chart in such a way that the point below which the stress is to be determined is located at the center of the chart.
- 4. Count the number of elements (M) of the chart enclosed by the plan of the loaded area.

The increase in the pressure at the point under consideration is given by

$$\Delta \sigma_z = (IV) q M$$

where IV = influence value q = pressure on the loaded area

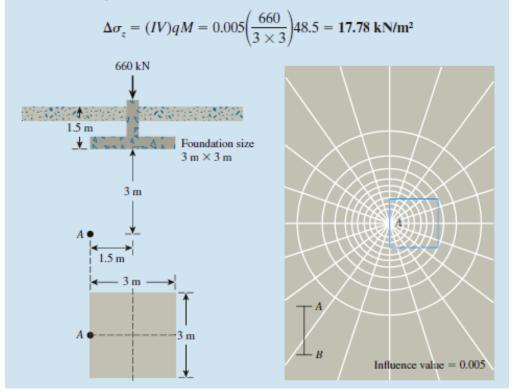
## **EXAMPLE 10.14**

#### Example 10.14

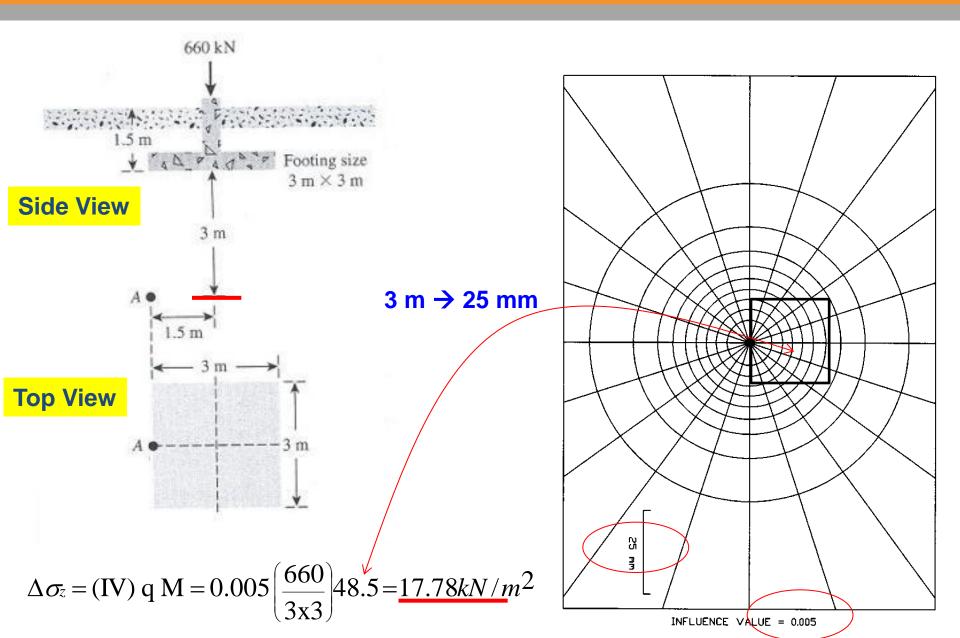
The cross section and plan of a column foundation are shown in Figure 10.29a. Find the increase in vertical stress produced by the column footing at point A.

#### Solution

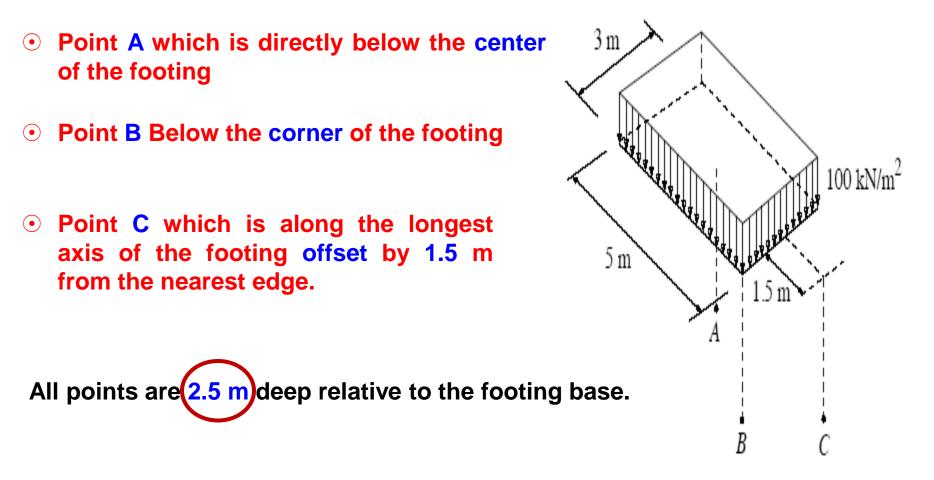
Point A is located at a depth 3 m below the bottom of the foundation. The plan of the square foundation has been replotted to a scale of  $\overline{AB} = 3$  m and placed on the influence chart (Figure 10.29b) in such a way that point A on the plan falls directly over the center of the chart. The number of elements inside the outline of the plan is about 48.5. Hence,



#### **EXAMPLE 10.13**

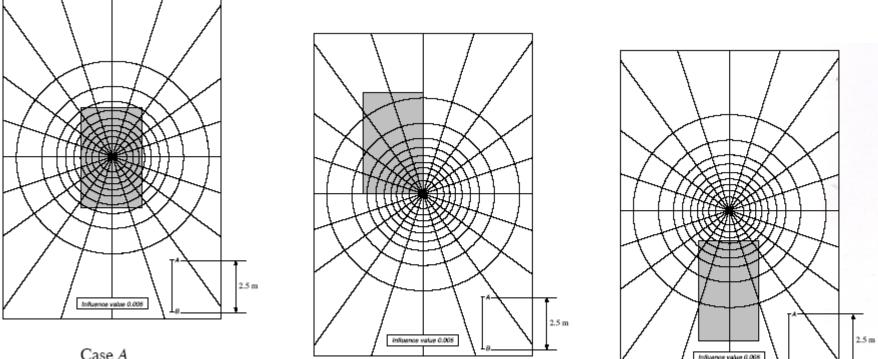


A rectangular footing is 3 m X 5 m and transmits a uniform load of 100 kPa into the soil mass. Compute the incremental vertical pressures at:



#### Solution

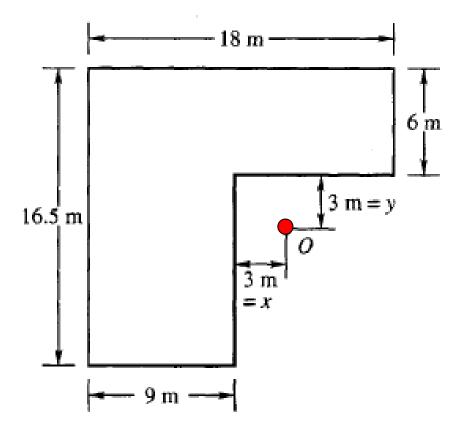
Let AB = 2.5 meters; sketch footing plans in the Newmark chart attached. For point A, use rectangle 1; n = 107.5 (verify);  $\Delta p = 107.5 \times 0.005 \times 100 = 53.8$  kN/m<sup>2</sup>. For point B, use rectangle 2; n = 42.0 (verify);  $\Delta p = 42.0 \times 0.005 \times 100 = 21.0$ kN/m<sup>2</sup>. For point C, use rectangle 3 (dashed); n = 20.2 (verify);  $\Delta p = 20.2 \times$  $0.005 \times 100 = 10.1 \text{ kN/m}^2$ .





Case C

A raft foundation of the size given below carries a uniformly distributed load of  $300 \text{ kN/m^3}$ . Estimate the vertical pressure at a depth 9 m below point O marked in the figure.



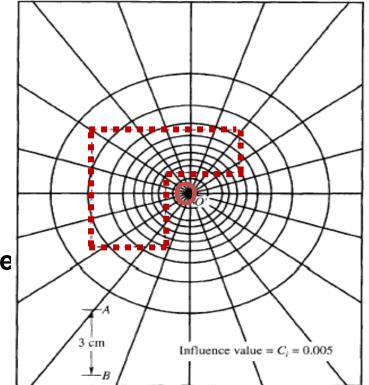
#### **Approach 2: Using Newmark's chart**

- The Depth at which  $\Delta\sigma$  required is 9 m
- From the Fig. across, the scale of the foundation plan is AB = 3 cm = 9 m or 1 cm = 3 m.
- Plot the loaded area at this scale.
- Superimpose the plan on the chart with point O coinciding with the center of the chart.
- Number of loaded blocks occupied by the plan, M = 62
- The vertical stress is given by:

 $\Delta \sigma = (IV) \times M \times q$ 

 $\Delta \sigma = 0.005 \times 62 \times 300 = 93 \text{ kN/m}^2$ 

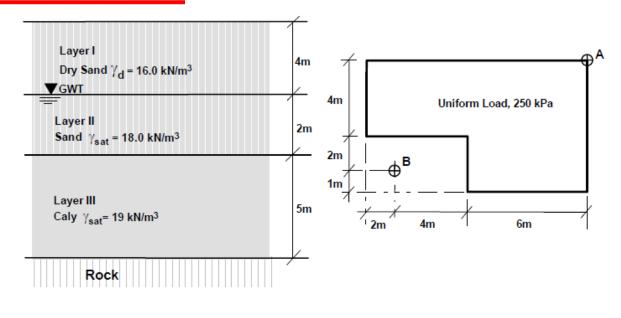
 $\Delta \sigma = 0.312 \times 300 = 93.6 \text{ kN/m}^2$  Approach 1: Superposition



a.For the soil profile shown in Fig. 2a, compute the pore water pressure and effective vertical stress at the mid-depth of layers I, II, and III.

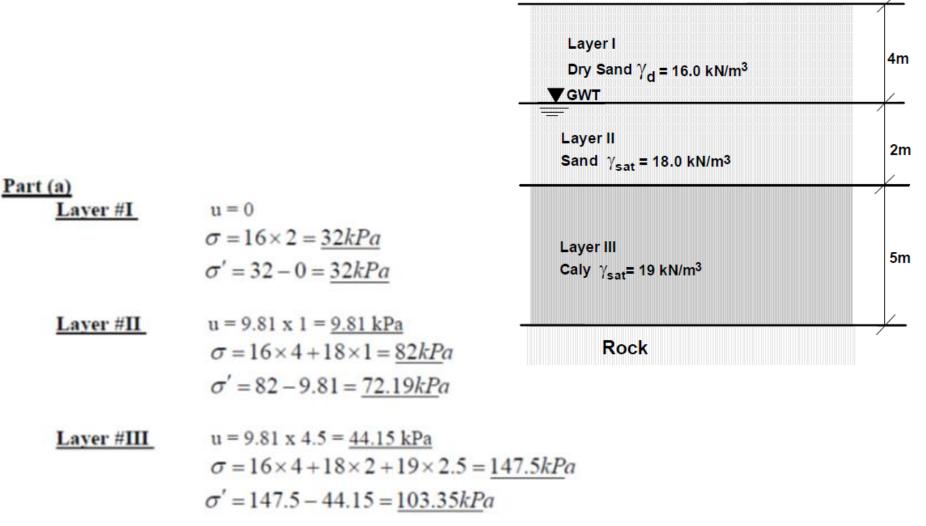
b.If on top of the soil profile shown in Fig. 2a, the loaded area shown in Fig. 2b is placed at the ground surface level. Compute: -

- The additional vertical stress, Δσz, due to the loaded area under point A at the middle of layer I (Use Fadum's method (m & n chart))
- ii. The additional vertical stress, Δσz, due to the loaded area, under point B at the middle of layer II. (Use Newmark's chart).









The additional vertical stress,  $\Delta \sigma_z$ , due to the loaded area under point A at the middle of layer I (Use Fadum's method (m & n chart))

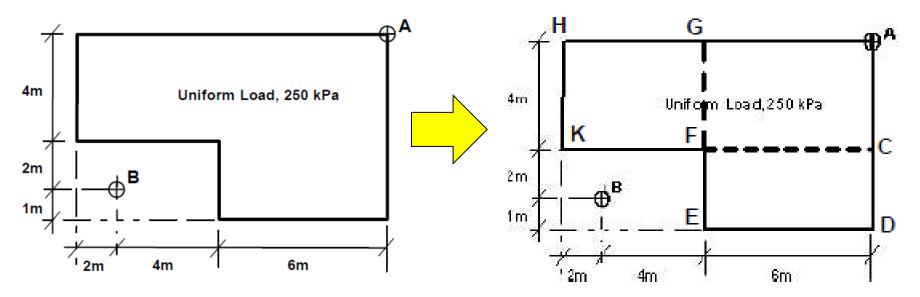
 Part (b)
 z= 2 m

 (i)
 Area CKHA  $m = 12/2 = 6; n = 4/2 = 2 \dots \rightarrow I = 0.24$  

 Area DEGA  $m = 7/2 = 3.5; n = 6/2 = 3 \dots \rightarrow I = 0.245$  

 Area CFGA  $m = 6/2 = 3; n = 4/2 = 2 \dots \rightarrow I = 0.238$ 

 $I = 0.24 + 0.245 - 0.238 = \underline{0.247}$  $\Delta \sigma = I \times q = 0.247 \times 250 = 61.75 \approx \underline{62kPa}$ 





The additional vertical stress,  $\Delta \sigma_z$ , due to the loaded area, under point B at the middle of layer II. (Use Newmark's chart).

